# NAVWEPS OP 2309 (VOLUME 4)

(THIRD REVISION

# AIM-9B GUIDED MISSILE (FORMERLY SIDEWINDER 1A)

# PREFLIGHT CHECKOUT AND LOADING AND UNLOADING MISSILE

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#### FOREWORD

Ordnance Pamphlet 2309 (Third Revision) describes the AIM-9B guided missile, gives the theory of operation, and covers handling, inspection, assembly, and stowage of components.

This publication consists of four volumes.

Volume 1-Description and Operation (CONFIDENTIAL)

Chapter 1-Description Chapter 2-Operation

Volume 2-Handling, Storage, and Fuze Assembly at Naval Weapons Stations

Chapter 1-Handling and Storage

Chapter 2-Fuze Assembly

Chapter 3-Records

Volume 3-Shipboard Handling, Inspection, Stowage, and Missile Assembly

Chapter 1-Shipping, Handling, and Stowage Requirements

Chapter 2-Component Inspection and Missile Assembly

Chapter 3-Records

Volume 4-Preflight Checkout and Loading and Unloading Missile

Chapter 1-Preflight Checkout Procedures

Chapter 2-Loading and Unloading Missile

Chapter 3-Records

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#### SAFETY SUMMARY

The following WARNINGS are repeated from the text for the protection of personnel.

#### WARNINGS

Before making a preflight test of the Aero 3A launcher, verify that all armament has been removed (both internal and external) from the aircraft. (Page 1-2)

Although the preflight jettison test is for AIM-9B missile circuits only, other aircraft jettison circuits may be energized during this test. Therefore, reaffirm that all ordnance (both internal and external) and other external stores have been removed from the aircraft. (Pages 1-4, 1-7, and 1-8)

Make sure that the firing system cannot be energized during missile loading operation. Never stand directly behind or in front of a missile that is being loaded onto a launcher. When a missile is loaded on a launcher, all unnecessary personnel should be at least 25 feet from the flank of the launcher and should not be behind the launcher. The loaded launcher is considered dangerous within 200 feet aft. With the exception of the final signal tone, no electrical tests are to be conducted while the missile is on the launcher. The safety pin shall remain in place until the aircraft is ready to catapult. For complete safety, aircraft power should be OFF during loading and unloading of missile. (Page 2-1)

When loading the missile on the LAU-7/A launcher, do not stand directly in front or behind the missile during loading operation. Stand clear of the launcher at all times after the missile

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is loaded on the launcher. Verify the safe condition of the launcher and aircraft before loading the missile on the aircraft by checking to (a) make certain that safety pin is installed properly, (b) verify that cockpit switches are OFF, (c) verify that aircraft engines are OFF and that auxiliary power is NOT connected to aircraft, and (d) make certain aircraft is grounded. (Page 2-2)

Failure to seat the detent properly when loading the missile on the launcher will result in either the missile not firing when the firing button is pressed or in loss of the missile during arrested landing or catapult launching. (Page 2-3)

Before removing a missile from the launcher make certain that the battery and aircraft armament switches are OFF and that the safety pin is in place on the launcher. (Page 2-4)

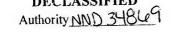
The following CAUTIONS are repeated from the text because if not strictly observed the effectiveness of the equipment or missile may be destroyed.

#### **CAUTIONS**

Visually check for improper positioning of both forward snubber cams. Return launchers with cracked or broken cams to an O and R facility for repair. Check for free movement when detent is raised. Check the unloading stirrup on the aft end of the Aero 3A launcher to be sure it is properly secured to the launcher and is not bent or deformed in any way. Check firing pin springs by depressing them; also check ceramic and teflon holders of firing pins. (Pages 1-2 and 1-6)

When testing the launcher with the AN/ASM-11 or AN/ASM-20 test set, do not leave test set selector switch in any test position for more than 5 minutes. (Pages 1-3, 1-6, and 1-7)

Never install a missile on the launcher with the NPA on the motor. Failure to remove the NPA will cause critical damage to the aircraft when the missile is launched. (Pages 2-1 and 2-3)



Never force a missile onto the launcher. Do not use excessive force or supplemental leverage in depressing the nose latch button. (Page 2-3)

The forward-receptacle dust cap must be kept in place on the launcher power supply receptacle at all times when the missile is not on the launcher. The upper receptacle dust cap must be on the aircraft-pylon receptacle on the launcher top side whenever the launcher is removed from the aircraft. (Page 2-3)

#### Chapter 1

#### PREFLIGHT CHECKOUT PROCEDURES

#### 1-1 INTRODUCTION

The Aero 3A launcher or LAU-7/A launcher can be used to fire AIM-9B missiles. The Aero 3A launcher fires AIM-9B missiles only, but the LAU-7/A fires AIM-9B, 9C, or 9D missiles.

1-1.1 AERO 3A LAUNCHER. The Aero 3A launcher operates from aircraft supplying single-phase power. The AN/ASM-11 or ASM-20 guided missile launcher test set is used to check the operability of the launcher and aircraft circuits. The 3/8-inch hex wrench is required to raise the detent of the launcher and to raise the launcher nose cover to connect the adapter connector to the launcher umbilical connector.

1-1.2 LAU-7/A LAUNCHER. The LAU-7/A launcher equipped with Power Supply PP-2581/A operates from single-phase aircraft power. When equipped with Power Supply PP-2315/A, it operates from either single-phase or three-phase aircraft power. This launcher and aircraft can also be checked with the ASM-11 or ASM-20 test set, but a launcher adapter (Part No. 10001-1517359, FSN VM-5935-885-9397-M558) must be installed when the ASM-20 test set is used. The 5/16-inch hex wrench is required to raise the detent of the launcher and is also the launcher safety pin. To connect the adapter to the umbilical connector, the nose cover of the launcher is pulled out or removed.

# 1-2 AN/ASM-11 GUIDED MISSILE LAUNCHER TEST SET



Figure 1-1. Guided Missile Launcher Test Set AN/ASM-11.

1-2.1 DESCRIPTION. This test set, figure 1-1, provides a go-no-go determination that the power supplied to the AIM-9B missile, both standby and firing, is within voltage tolerance that the missile firing sequence is operating satisfactorily, and that the firing circuits are safe. In addition, it may be used to show that the aircraft-launcher circuits are operational and that the jettison circuits function satisfactorily.

The AN/ASM-ll test set is contained within a waterproof and shock-resistant carrying case; total weight

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of the set is 25 pounds, including the carrying case. The front panel of the test set is equipped with two handles that afford crash protection as well as handholds in moving the unit from one launcher station to another. The front panel display consists of a pilot light, an indicator meter, a 20-position selector switch, and an electrical connector.

A special purpose electrical cable assembly is furnished as part of the test set. The cable assembly provides the required electrical connections between the test set and the missile launcher or aircraft circuits being tested.

An adapter connector is furnished which is used to adapt the test set cable assembly to the aircraft circuits when the launcher has been removed from the aircraft and disconnected from the aircraft circuitry.

1-2.2 PREFLIGHT TEST OF AERO 3A LAUNCHER. When using the AN/ASM-11 test set to check Aero 3A launcher functions, proceed as follows:

> CAUTION: Visually check for improper positioning of both forward snubber cams. Return launchers with cracked or broken cams to an O and R facility for repair. Check for free movement when detent is raised. Check the unloading stirrup on the aft end of the Aero 3A launcher to be sure it is properly secured to the launcher and is not bent or deformed in any way. Check firing pin springs by depressing them; also check ceramic or teflon holders of firing pins.

- l. Place test set in operating position and observe if the knife edge pointer on the meter is on the extreme left edge of the BLACK area. If this is not true, refer to NAVWEPS 16-30ASM11-1 for calibration instructions.
- 2. Connect connector Pl02 of cable assembly to connector Jl01 on front panel of test set.

#### WARNING

Before proceeding, verify that all armament has been removed (both internal and external) from the aircraft.

- 3. Install safety pin in launcher.
- 4. Raise detent of launcher with hex wrench and insert lug handle of cable assembly into position on launcher rail, figure 1-2. (The arrow on the lug handle should point in the aircraft forward direction.)
- 5. Raise launcher nose cover, using hex wrench, and connect Pl03 to launcher connector (AN-3102E-22-14P), figure 1-3.
- 6. Energize aircraft circuitry by use of an auxiliary power unit.



Figure 1-2. ASM-11 Test Set Connected to Aero 3A Launcher.



Figure 1-3. Lug Handle Inserted and Cable Assembly Connected.



Figure 1-4. Launcher Test Set Connected and Ready for Launcher Checkout.

CAUTION: Do not leave test set in any test position more than 5 minutes.

7. Starting with position 1, turn switch knob through first 10 positions, stopping at position 10. Be sure that knife edge pointer on meter dial is

in the YELLOW area for each of the 10 positions, figure 1-4.

- 8. With switch knob on position 10, listen for a 400-cycle tone in the pilot's earphones. If the tone is not heard, be sure that the proper cockpit switches are turned on and that the volume is turned up.
- 9. Turn MASTER ARMAMENT and SAFETY OVERRIDE switches ON.
- 10. Depress firing switch and hold is down, then
- a. Turn switch knob through positions 11, 12, and 13. (The knife edge pointer should read in the RED area.)
- b. Next, turn switch knob to position 14. (The knife edge pointer should read in the BLACK area.)
- c. Pull safety pin from launcher (At position 14, the pointer should now be in the RED area.)
- 11. Release firing switch, and turn MASTER ARMAMENT switch OFF.
- 12. With safety pin removed, turn switch knob to position 15. (The knife edge pointer should read ZERO (BLACK area), and the pilot light should go out.)
- 13. With safety pin still removed, turn switch knob to positions 16, 17, and 18. (The knife edge pointer should read in the BLACK area for all three positions.)
- 14. Turn switch knob to position 19. (The knife edge pointer should read in the YELLOW area.)
- 15. Turn switch knob to OFF position, and reinsert safety pin in launcher.
- 1-2.3 PREFLIGHT LAUNCHER JETTISON TEST. This test is performed on the jettison circuitry of the Aero 3A and does not confirm

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any circuits related to the external stores or tanks.

#### WARNING

Although the following jettison test is for AIM-9B missile circuits only, other aircraft jettison circuits may be energized during this test. Therefore, reaffirm that all ordnance (both internal and external) and other external stores have been removed from the aircraft.

#### Proceed as follows:

- 1. Energize aircraft circuitry.
- 2. Turn test set switch knob to position 14.
- 3. Perform jettison procedures prescribed for type of aircraft being tested. (The knife edge pointer should be in the RED area.)
  - 4. De-energize jettison circuitry.
- 5. Turn switch knob through positions 16, 17, and 18. (The knife edge pointer should be in the BLACK area for all three positions.)
- 6. Turn off, or remove, power; and disconnect test set.
- 1-2.4 PREFLIGHT AIRCRAFT CIRCUITRY TEST. The following test is made on aircraft when NO missile launchers are on the aircraft.

NOTE: This test cannot be made on an aircraft that does not have a connector MS 3106A-20-27S at or near the launcher station.

- 1. Connect P102 of cable assembly to J101 of test set.
- 2. Connect P105 of adapter to connector MS 3106A-20-27S located

in or on aircraft wing or fuselage, or at the bottom of launcher pylon.

- 3. Connect Pl03 of cable assembly to Pl04 of adapter.
- 4. Energize aircraft circuity, using an auxiliary power unit.
- 5. Turn MASTER ARMAMENT and SAFETY OVERRIDE switches ON.
- 6. Turn switch knob to positions 2, 3, and 10. (The knife edge pointer should read in the YELLOW area for all three positions, and a 400-cycle tone should be heard in the pilot's earphones.)
  - 7. Turn switch knob to position 12.
- 8. Depress firing switch. (The knife edge pointer should read in the RED area.)
- 9. Turn power OFF, and disconnect test set.

If the test set at any time, indicates that the aircraft or launcher circuit is not functioning properly, all minor causes for negative results should be eliminated before labeling the aircraft, launcher, or test set as defective. (See paragraph 1-2.5.)

- 1-2.5 MINOR CAUSES OF NEGATIVE RESULTS. The following information should serve as a guide in establishing the point of existing trouble should the test set, at any time, indicate that the aircraft, or launcher, circuits are not functioning properly:
- 1. Check test set selector switch to be sure that it is on the correct position.
- 2. Check connector at J101 and all other connections to be sure that they are correct and electrically secure.
- 3. Check auxiliary power unit for proper output.

- 4. Check position of switches in aircraft.
- 5. If trouble still exists, connect test set to another launcher or aircraft circuit. If the same failure occurs, obtain a new test set and check the troubled position again. Should the same condition remain, change back to the original test set; label the aircraft circuit or launcher as defective; and mark the aircraft circuit or launcher for repair.
- 6. Should the test set prove to be out of order, mark the test set for repair.
- 1-2.6 PREFLIGHT TEST OF LAU-7/A LAUNCHER. When using the AN/ASM-11 test set to check the LAU-7/A launcher functions, the procedures are the same as those given in paragraphs 1-2.2 through 1-2.5, except for the following change to paragraph 1-2.2, step 10:

NOTE: In performing these tests, the missile-to-launcher adapter (Part No. 10001-1517359, FSN. VM-5935-885-9397-M558) is required and these instructions apply only to LAU-7/A launchers in which the HVAR receptacle has been removed and the wiring modified.

- 10. Depress the firing switch and hold it down, then
- a. Turn switch knob through positions 11 and 12. (The knife edge pointer will read in the RED area.)
- b. Next, turn switch knob through positions 13 and 14. (The knife edge pointer will read in the BLACK area.)
- c. Pull safety pin from launcher. (At positions 13 and 14, the pointer will now read in the RED area.)

# 1-3 AN/ASM-20 GUIDED MISSILE LAUNCHER TEST SET

The Aero 3A and LAU-7/A launcher; can also be tested for safe operability with the AN/ASM-20 test set, figure 1-5. When checking the LAU-7/A launcher, a special adapter (Part No. 10001-1517359, FSN VM-5935-885-9397-M558) is required in addition to the adapter connectors furnished with the test set.

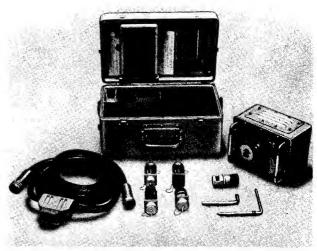


Figure 1-5. Guided Missile Launcher Test Set AN/ASM-20.

1-3.1 DESCRIPTION. The components of the ASM-20 test set are as follows: (1) transit case, (2) test set, (3) cable assembly, (4) four adapters, (5) pressure gage, and (6) hex wrenches. A pilot's headset, compatible with the aircraft being tested, and an auxiliary power source to energize the aircraft circuitry must be available. One of the adapter connectors (U-216/U) and the pressure gage are not required for preflight testing of AIM-9B missiles. Instruction cards covering all three types of missiles (AIM-9B as well as AIM-9C and AIM-9D) and power supplies required, along with a copy of the handbook (NAVWEPS 16-30ASM-20-1), are included with each test set.

1-3.2 PREFLIGHT LAUNCHER TESTS When using the AN/ASM-20 test set

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to check the Aero 3A or LAU-7/A launcher functions, proceed as follows:

CAUTION: Visually check for improper positioning of both forward snubber cams. Return launchers with cracked or broken cams to an O and R facility for repair. Check for free movement when detent is raised. Check the unloading stirrup on the aft end of the Aero 3A launcher to be sure it is properly secured and is not bent or deformed in any way. Check firing pin springs; also check ceramic and teflon holders of firing pins.

- l. Place test set in operating position and observe if the knife edge pointer on the meter is on the extreme left edge of the BLACK area. If this is not true, refer to NAVWEPS 16-30ASM20-1-1 for calibration instructions.
- 2. Remove all electro-explosive device ordnance from aircraft.
- 3. Insert lug handle of test set cable assembly into launcher.
- 4. Connect test set cable assembly with appropriate adapter to launcher umbilical connector:
- a. For Aero \$A launcher use Adapter U-213/U.
- b. For LAU-7/A launcher use Adapter U-213/U with Adapter Part No. 10001-1517359 (FSN VM-5935-885-9397-M558).
- 5. Place aircraft ARMAMENT SELECTOR switch on SIDEWINDER position.
- 6. Place aircraft STATION SELECTOR switch on station to be tested.
- 7. Place test set MISSILE ID switch on 1A position.

8. Apply external auxiliary power to aircraft.

CAUTION: Do not leave test set selector switch in any test position for more than 5 minutes.

#### NOTE

- a. Pilot light will glow in all positions unless other-wise noted.
- b. Test only on 1A test position (A-on code bars).
- c. Leave selector switch on position 6 for at least 1 minute prior to continuing tests
- d. Before repeating any test outlined below turn all aircraft switches OFF.
- 9. a. Aero 3A launcher: Rotate switch to positions 6 through 15. (Meter should read in YELLOW area for each position. When switch is on position 13, a 400-cycle tone should be heard in the pilot's headset. Some aircraft require that the communication system be ON.)
- b. LAU-7/A launcher: Rotate switch to positions 6 through 13 and 15. (Meter should read in YELLOW area for each position. When switch is on position 13, a 400-cycle tone should be heard in the pilot's headset. Some aircraft require that the communication system be ON.)

  Next, rotate switch to position 14. (Meter should read in BLACK area.)
- 10. Steps 11 through 13 below are omitted if firing circuits are not tested.
- 11. Install launcher safety pin if not already installed.
- 12. Turn ON aircraft MASTER ARMAMENT and ARMAMENT SAFETY OVERRIDE switches.
- 13. Hold firing switch down and rotate switch through the following positions:

- a. Aero 3A launcher: Positions 16, 17, 19, 21, and 22. Positions 16, 17, and 19 should read in RED area. Positions 21 and 22 should read in BLACK area. Pull launcher safety pin. Repeat switch positions 21 and 22. Meter should read in RED area. Release firing switch.
- b. LAU-7/A launcher: Positions 16 and 17 should read in RED area. Release firing switch. Rotate to positions 19, 21, and 22. Meter should read in BLACK area. Pull launcher safety pin. Repeat switch positions 19, 21, and 22. Meter should read in RED area.
- 14. Turn aircraft MASTER ARMA-MENT switch OFF.
- 15. Steps 16 through 18 below are omitted if jettison circuit is not tested.

#### WARNING

Although the following jettison test is for AIM-9B missile circuits only, other aircraft jettison circuits may be energized during this test. Therefore, reaffirm that all ordnance (both internal and external) and other external stores have been removed from the aircraft.

- 16. Rotate switch to position 21.
- 17. Perform jettison test as prescribed for aircraft being tested.
  (Meter should read in RED area.)
- 18. De-energize aircraft jettison circuit.
- 19. Turn aircraft MASTER ARMA-MENT switch ON.
- 20. Rotate switch to position 23. (Meter should read in BLACK area, and pilot light should not glow.)
- 21. With safety pin removed, rotate switch to positions 24, 26, and

- 28. (Meter should read in BLACK area.)
- 22. Rotate switch to position 29. (Meter should read in YELLOW area.)
  - 23. Rotate switch to OFF position.
- 24. Turn aircraft MASTER ARMA-MENT switch OFF and remove external auxiliary power.
- 25. Replace launcher safety pin before loading missile.
- 1-3.3 PREFLIGHT AIRCRAFT TESTS
  Proceed as follows:
- 1. Remove all electro-explosive device ordnance from aircraft.
- 2. Connect test set cable assembly with appropriate adapter to pylon connector.
- a. For aircraft circuits designed for the Aero 3A launcher use Adapter U-214/U.
- b. For aircraft circuits designed for the LAU-7/A launcher use Adapter U-215/U.
- 3. Place ARMAMENT SELECTOR switch on SIDEWINDER position.
- 4. Place aircraft STATION SELECTOR switch on station to be tested.
- 5. Apply external auxiliary power to aircraft.
- 6. Place MASTER ARMAMENT and ARMAMENT SAFETY OVER-RIDE switches in the ON positions.
- 7. Place MISSILE ID switch in 1A position.

CAUTION: Do not leave test set selector switch in any test position for more than 5 minutes.

#### NOTE

a. Pilot light will glow in

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all positions unless otherwise noted.

- b. Test only on 1A test position (A-on code bars).
- 8. Rotate switch to positions 30 and 32. (Meter should read in YELLOW area for each position.)
- 9. Rotate switch to position 33. Depress firing switch. (Meter should read in YELLOW area.) Release firing switch.
  - 10. To test missile signal:
    - a. Aero 3A launcher:

Rotate switch to position 34. (an audible signal should be heard in the pilot's headset.)

- b. LAU-7/A launcher:
  - (1) For aircraft circuitry designed to supply input power to Power Supply PP-2581/A, rotate switch to position 34. (An audible signal should be heard on the pilot's headset.)
  - (2) For aircraft single-phase power used as input power for Power Supply PP-2315/A, rotate switch to positions 34, 36, and 37. (An audible signal should be heard in the pilot's headset for positions 34 and 36.)

(3) For aircraft three-phase power used as input power for Power Supply PP-2315/A, rotate switch to positions 34, 35, and 36. (An audible signal should be heard in the pilot's headset for each position.)

#### WARNING

Although the following jettison test is for AIM-9B missile circuits only, other aircraft jettison circuits may be energized during this test. Therefore, reaffirm that all ordnance (both internal and external) and other external stores have been removed from the aircraft.

- 11. Rotate switch to position 38. Perform the jettison test as prescribed for aircraft being tested. (Meter should read in RED area.)
- 12. De-energize the aircraft armament and jettison circuits.
- 13. Rotate switch to position 40. (Meter should read in YELLOW area.)
- 14. Rotate switch to position 42. (Meter should read in BLACK area, and pilot light should <u>not</u> glow.)
  - 15. Rotate switch to OFF position.
- 16. Remove external auxiliary power.

#### Chapter 2

#### LOADING AND UNLOADING MISSILE

#### 2-1 LOADING MISSILE ON LAUNCHER

If possible, the aircraft should be pointed away from other aircraft, personnel, and structures. A threeman team is required to load the missile on the launcher. The procedures are given in the following paragraphs.

#### WARNING

Make sure that the firing system cannot be energized during loading operation. Never stand directly behind or in front of a missile that is being loaded onto a launcher. When a missile is loaded on a launcher, all unnecessary personnel should be at least 25 feet from the flank of the launcher and should not be behind the launcher. The loaded launcher is considered dangerous within 200 feet aft. With the exception of the final signal tone, no electrical tests are to be conducted while the missile is on the launcher. The safety pin shall remain in place until the aircraft is ready to catapult. For complete safety, aircraft power should be OFF during loading and unloading of missile.

2-1.1 LOADING MISSILE ON AERO 3A LAUNCHER. The procedure for loading the AIM-9B missile on the Aero 3A launcher is as follows:

CAUTION: Never install a missile on the launcher with the NPA on the motor. Failure to remove the NPA will cause critical damage to the aircraft when the missile is launched.

- 1. Remove NPA from motor.
- 2. Check that dust cover is on power supply receptacle when umbilical is disconnected, figure 2-1.

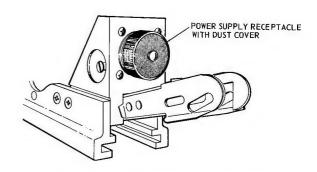


Figure 2-1. Dust Cover on Power Supply Receptacle of Aero 3A Launcher.

NOTE: All Aero 3A launchers have a dust cover protecting the power supply receptacle. This dust cover shall always be in place when the umbilical is not connected to the

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power supply. An unprotected power supply receptacle exposed to weather can result in an electrical short in the power supply and cause the launcher to fail.

3. Make certain that aircraft battery and MASTER ARMAMENT switches are OFF and that launcher safety pin is in place, figure 2-2.



Figure 2-2. Launcher on Aircraft With Safety
Pin in Place.



Figure 2-3. AIM-9B Missile Being Loaded Aboard Aircraft.

- 4. Bring missile into position just outside three loading slots; move missile to engage it in slots.
- 5. Raise aft lug of detent by rotating it with 3/8-inch hex wrench.
- 6. Push missile forward about 3 inches, figure 2-3, until forward lug engages detent.
  - 7. Remove wrench from launcher.
- 8. Unlatch front cover on launcher, remove dust cap, and connect umbilical cord and umbilical disconnect mechanism. Relatch and secure cover.
- 9. Make certain that snubber cams are in locked position and that safety pin is in place.
- 10. Just before aircraft moves forward remove protective covers from G&C section and influence fuze, and perform simplified missile checkout given in paragraph 2-2.
- 2-1.2 LOADING MISSILE ON LAU-7/A LAUNCHER. When possible, the aircraft should be pointed away from other aircraft, personnel, and structures. Proceed as follows:

#### WARNING

Do not stand directly in front of or behind the missile during loading operation. Stand clear of the launcher at all times after the missile is loaded on the launcher. Verify the safe condition of the launcher and aircraft before loading the missile on the aircraft by checking each of the following conditions:

a. Make certain that safety pin is installed properly.

b. Verify that cockpit switches are OFF.

c. Verify that aircraft engines are OFF and that auxiliary power is NOT connected to aircraft.

d. Make certain aircraft is grounded.

CAUTION: Never install a missile on the launcher with the NPA on the motor. Failure to remove the NPA will cause critical damage to the aircraft when the missile is launched.

- 1. Remove NPA from motor.
- 2. Insert motor hangers into launcher loading slots.
- 3. Slide missile forward until it hits aft detent lug.
- 4. Raise detent by rotating detent wrench-safety pin, and slide missile forward until front edge of hanger block engages front detent lug.
- 5. Lower detent by releasing, but not removing, detent wrench-safety pin.

#### WARNING

Failure to seat the detent properly will result in either the missile not firing when the firing button is pressed or in loss of the missile during arrested landing or catapult launching.

6. Check carefully to see that the missile is properly loaded and that the detent is not resting on the firing button of the missile. Two improper conditions are illustrated in figure 2-4.

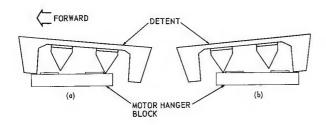


Figure 2-4. Improper Loading Conditions.

a. When the rear leg of the launcher detent is raised with the detent wrench, and the missile is shoved forward too far and with too much force, the front of the motor hanger block will cause the front leg of the launcher detent to rise and come to rest on top of the hanger block (see figure 2-4a). This prevents prope engagement of the launcher detent and prevents the striker points from contacting the firing buttons.

b. Once the launcher detent has been lowered and has engaged the motor hanger block, manual rotation of the detent wrench will force the front leg of the detent to move the missile aft, at the same time raising the rear leg of the detent. When the detent wrench handle is released, the rear leg of the detent will rest on the motor hanger block (see figure 2-4b). This causes disengagement of the detent and prevents the striker points from contacting the firing buttons.

CAUTION: Never force a missile onto the launcher. Do not use excessive force or supplemental leverage in depressing the nose latch button.

- 7. Depress nose latch button and slide nose fairing forward.
- 8. Check umbilical hook for damage, then attach umbilical hook to missile umbilical block by pushing hook down until it snaps into place over the umbilical-block pin.

CAUTION: The forward-receptacle dust cap must be kept in place on the launcher power supply receptacle at all times when the missile is not on the launcher. The upper receptacle dust cap must be

#### THIRD REVISION

on the aircraft-pylon receptacle on the launcher top side whenever the launcher is removed from the aircraft.

9. Remove forward-receptacle dust cap shown in figure 2-5, and, using an adapter furnished specifically for the AIM-9B missile, connect missile umbilical cable to launcher power supply receptacle.



Figure 2-5. Forward-Receptacle Dust Cap

To Be Removed.

10. Push nose fairing home.

11. Just before aircraft moves forward, remove protective covers from G&C section and influence fuze, and perform the simplified missile checkout given in paragraph 2-2.

NOTE: The safety pin IS NOT removed until after the missile has been given the simplified checkout and the aircraft is ready to taxi forward.

If, for some reason, the detent wrench is rotated after the missile is loaded on the aircraft, corrective measures (appendix A) are recommended.

#### 2-2 SIMPLIFIED MISSILE CHECKOUT

With the assembled missile on the launcher and with protective covers and NPA removed, but with safety pin still in place on the launcher, check out the missile as follows:

- 1. Apply power through the aircraft circuits.
- 2. Inspect to see that gyro in seeker is rotating. If it is not rotating, check missile umbilical connector to see that it is fully engaged in the launcher; also check fuses and switches in the aircraft. If the gyro still fails to rotate, try to start it by moving any magnetic object (such as a bar magnet or a steel tool) clockwise around the head behind the glass dome. If this does not start the gyro, reject the unit.
- 3. With the gyro rotating, check pilot's tone by passing a portable infrared source, such as a standard two-cell flashlight that has glass lens, not plastic, about 15 feet from the front of the seeker head. If the pilot's head-phone is functioning and no audible signal is heard, reject the unit.
- 4. When the missile passes this simplified checkout and the aircraft is ready to taxi forward, pull the launcher safety pin.

#### 2-3 UNLOADING MISSILES FROM LAUNCHER

2-3.1 UNLOADING MALFUNCTIONING MISSILE. A malfunctioning missile is one which does not leave the launcher in the normal manner.

#### WARNING

Before removing a missile from the launcher, make certain that the battery and aircraft armament switches are OFF and that the safety pin is in place on the launcher.

#### Proceed as follows:

- 1. Make certain that battery and aircraft armament switches are OFF.
  - 2. Place safety pin in launcher.
  - 3. Move aircraft to a safe position.
- 4. If a missile which malfunctioned is aboard, wait 10 minutes from the time that the firing button was last depressed before unloading the missile.
- 5. Unload missile as indicated in paragraph 2-3.2.

NOTE: BUWEPS INST 8020.6B is to be followed in reporting malfunctions.

- 2-3.2 UNLOADING LIVE, CAPTIVE-FLIGHT, OR TRAINING MISSILES. The procedure for unloading normal missiles is as follows:
- 1. Make certain that battery and aircraft switches are OFF.
  - 2. Place safety pin in launcher.
- 3. Raise detent, using appropriate launcher wrench, to allow the missile to be moved backward to the loading slots.

- 4. Use care in handling missile so that plastic strip on influence fuze is not damaged (scratched, punctured, or marred) in any way.
- 5. Place protective covers on G&C section and influence fuze before removing the missile from launcher.
- 6. Remove missile from launcher, and place NPA on motor.
- 7. Replace shorting plug on umbilical plug.
- 8. Examine firing button plastic caps on HERO SAFE motors. If damaged, general HERO safety precaution shall be observed.
- 9. Move missile to assembly area where components are disassembled, inspected, and segregated or disposed of as described in volume 3, paragrap 2-6.

NOTE: It is recommended that the umbilical cable be secured to the body of the G&C section at all times, except when in use or when testing

#### Chapter 3

#### RECORDS

ne Air-Launched Guided Missile pon System Performance Data orting Program has been estabed to collect data from representa-Fleet squadrons under day-to-day ating conditions. Firings, misgs, and malfunctions of the AIM-9B sile are to be reported in accor-

dance with BUWEPS INST 8810.2 of 7 December 1964. Forms to be used in this program are reproduced as figures 3-1 (4 parts), 3-2 (4 parts), 3-3, and 3-4. Check lists for recording steps of procedure in aircraft checkout and in loading and unloading missiles are given in appendix B.

|        |                | NORK,   | CALIF   | QRNIA   | 91720    | _     |                      |           |            | _    | _      |           | -    |                   | ON GROUP.           |
|--------|----------------|---------|---------|---------|----------|-------|----------------------|-----------|------------|------|--------|-----------|------|-------------------|---------------------|
| -      | RCRAF          |         |         | LOT     |          | _     |                      |           | DATA       | _    |        |           | 17   | AP                |                     |
|        | RCRAI          | 1       | PI      | LUI     | $\dashv$ | _     | CALL                 | $\dashv$  |            |      |        |           | 1    |                   |                     |
|        |                |         |         |         | - 1      |       |                      |           |            | 11   | FF     |           | F    | REQ. CHA          | NNELS               |
|        |                |         |         |         |          |       |                      |           | 1,         |      | 101.   |           | ٦    |                   |                     |
|        |                | _       |         |         | _        | _     |                      | $\exists$ |            |      |        |           | 1    |                   |                     |
|        |                |         |         |         |          |       |                      | -         | AL TIME TE | R    |        |           | P    | RESSURE           | RATID               |
|        |                | +       |         |         | -        | _     |                      | $\dashv$  | WEATHER    | BR   | IĒ F   |           |      |                   |                     |
|        |                |         |         |         |          |       |                      |           |            |      |        |           |      |                   |                     |
|        |                |         |         | _       | _        | DIV   | ERT FIE              | LDI       | NFORM      | ATI  | ION    |           | _    |                   |                     |
|        | FIEL           | D       |         | BEA     | RING     |       | DIS                  | TAN       | CE         |      | T      | ACAN      |      |                   | сомм.               |
|        |                |         |         |         |          |       |                      |           |            |      |        |           |      |                   |                     |
|        |                |         | +-      |         |          |       |                      |           |            |      |        |           |      |                   |                     |
|        |                |         |         |         |          |       |                      |           |            |      |        |           |      |                   |                     |
|        |                |         | +       |         |          |       |                      |           | -          |      |        |           |      |                   |                     |
|        |                |         |         |         |          |       |                      |           |            |      |        |           |      |                   |                     |
| FL     | IGHT           | TIME    | +       |         |          | _     | MAF                  | RSHA      | ALING II   | NF(  | DRM    | ATION     | _    |                   |                     |
| TAKE   | DFF            |         | 1       |         | T        | ВЕ    | ARING                | T         | DISTAN     | _    | T      | AL T      | _    | DE                | E. A. C.            |
|        |                |         |         |         | T        |       |                      |           |            | _    | $\top$ |           |      |                   |                     |
| LAND   | NNG            |         | MAR     | RSHAL   | $\perp$  |       |                      |           |            |      | _      |           |      |                   |                     |
|        |                |         |         | RGENO   | Y        |       |                      |           |            |      | - 1    |           |      |                   |                     |
| BRIE   | FING N         | OTES:   | 1 ***   | JIIAL   |          | _     |                      | _         |            | _    |        |           | _    |                   |                     |
|        |                |         |         |         |          |       |                      |           |            |      |        |           |      |                   |                     |
|        |                |         |         |         |          |       |                      |           |            |      |        |           |      |                   |                     |
|        |                |         |         |         |          |       |                      |           |            |      |        |           |      |                   |                     |
|        |                |         |         |         |          |       |                      | ROP       | DATA       |      |        |           |      |                   |                     |
| NAY    | Y/<br>IE SQDN. | 2. SHII | P/STATE | DH (TA  | KEOFF    | LO    | CATION)              |           | 3. DATE    | Mon  | nth, D | lay, Year | , 1  | NC.               | 5. SYLLABUS<br>CODE |
| TYP    | EEV            | 7. AIR  | DACT    | To A    | C BUNE   | _     | 0.4/5.504            | FIC       | 10, FLIGH  |      | h. cc  | NITROLL   |      | CONTRA            | V 13 BIT CHECK      |
| Cod    | (e)            | TYP     | E       | ° (L    | sar 5 di | gits. | 9. A/C CON<br>(Code) | rig.      | GEAR/Ce    | efa) | CAL    | L SIGN    | 1    | TYPE (Code        | (Code)              |
|        | N INFO         | RMATIC  | N (CO   | DEI     | _        | 17.   | P ER SONNE           | L/3       | nitio(a)   | ke   | FLIG   | HT TIME   | 19.5 | P IN/CW O         | N 20. A/C LANDIN    |
| 4155HC |                |         | _       | 16. REA | SON      | -     | PILOT                |           | RIO        | 1    | -      | TENTHS    | _    | TIME<br>IRS TENTH | COND. (Code)        |
| _      | <b>OPOSED</b>  |         |         |         |          |       |                      |           |            |      |        |           |      |                   |                     |
| 4. PRO | 2ND            | 157     | 2NO     | 157     | 2NO      |       |                      |           |            | 1    |        |           |      |                   |                     |

|   | C        | ONFID  |  |         | filled in                                     |          |  |
|---|----------|--------|--|---------|---|----------|--|
| 21. RUN NUMBER  | 1        | 2      | 3  | 1 ERCEI | 5   | 6        | INSTRUCTIONS AND CODING  |
| 22. FIGHTER   | -        |        | 3  | -       | ,   |          | REPORT IN FEET   |
| ALTITUDE  |          |        |  |         |   |          | (TO NEAREST HUNDRED)   |
| 23. FIGHTER<br>MACH   |          |        | •  | •       |   |          | REPORT TO NEAREST HUNOREOTH<br>(EXAMPLE: 1.21 OR 0.85)   |
| 24. BUGEY<br>ALTITUDE   |          |        |  |         |   |          | REPORT IN FEET.<br>(TO NEAREST NUNORED)  |
| 25. BOGEY MACH  |          |        |  |         |   |          | REPORT TO NEAREST HUNDREDTH<br>(EXAMPLE: 1,27 OR 0,85)<br>REPORT V <sub>C</sub> IF MACN IS UNKNOWN<br>(EXAMPLE: 1200 KNOTS)  |
| 26. BOGEY TRACK<br>CROSSING ANGLE   |          |        |  |         |   |          | REPDRY IN DEGREES AT DETECTION<br>EXAMPLE: TAIL ON - 0°<br>NEAD DN - 180°  |
| 27 DE TECTION<br>OPPORTUNITY<br>RANGE   |          |        |  |         |   |          | REPORT IN NAUTICAL MILES.  |
| 28. BOGEY<br>AZIMUTH ANGLE  |          |        |  |         |   |          | REPORT IN DEGREES FROM BORESIGHT<br>AT DETECTION   |
| 29. DETECTION<br>RANGE  |          |        |  |         |   |          | REPORT IN NAUTICAL MILES   |
| 30. LOCK-ON<br>RANGE  |          |        |  |         |   |          | REPORT IN NAUTICAL MILES   |
| 31. TYPE OF<br>ATTACK   |          |        |  |         |   |          | A - SPARROW II) E - IDENTIFICATION<br>B - SIDE WINDER<br>C - SP III 'SW REATTACK   |
| 32. WEATHER AT<br>FIGHTER   |          |        |  |         |   |          | A · CAVU<br>B · CLDUDS<br>C · RAIN   |
| 33. WEATHER AT<br>BOGEY   |          |        |  |         |   |          | F - SNOW<br>F - THUNDERSTORMS<br>X - OTHER (EXPLAIN)   |
| 34. CONTROLLER<br>NUMBER  |          |        |  |         |   |          | GIVE APPROPRIATE IDENTIFICATION<br>NUMBER ASSIGNED TO CONTROLLER<br>A-ND CONTROLLER  |
| 35. ASSESSMENT<br>OF CONTROLLER   |          |        |  |         |   |          | A - NO ERROR<br>D - BEARING ERR - OEGREES (EX-DIS)<br>F - ALTITUDE ERR - FEET (EX: F1000)<br>M - RANGE ERR - MILES (EX: MZ)  |
| 36. RADAR MODE  |          |        |  |         |   |          | A-MAPPING E-IR DETECTION<br>B-BST F-RADAR<br>C-NUTATE 1-BAR X-OTHER (EXPLAIN)  |
| 37. SWEEP<br>DISPLAY  |          |        |  |         |   |          | A - NARROW E - EXPANDED B - WIDE X - OTHER (EXPLAIN) C - MANUAL SPOTLIGHT  |
| 38. BOGEY TYPE  |          |        |  |         |   |          | USE CODES FROM AAMREP CODE SHEET   |
| 39. BOGEY<br>CHARACTERISTICS  |          |        |  |         |   |          | A - AUGMENTED, NOT MANEUVERED<br>B - AUGMENTED, MANEUVERED<br>C - UNAUGMENTED, NOT MANEUVERED<br>E - UNAUGMENTED, MANEUVERED |
| 40. TRACKING  |          |        |  |         |   |          | A - AUTO MANUAL E - VISUAL<br>B - MANUAL MANUAL<br>C - NORMAL X - DTHER (EXPLAIN)  |
| 4). CLUTTER   |          |        |  |         |   |          | B-HEAVY<br>C-OVERRIDE  |
| 42 POLARIZATION   |          |        |  |         |   |          | A-LINEAR<br>B-CIRCULAR I<br>C-CIRCULAR 2   |
| 43. WAS CM<br>ENCOUNTEREO   |          |        |  |         |   |          | A-YES B-ND IF YES, FILL OUT SUPPLEMENT CM CARC AND ATTACH TO THIS FORM.  |
| 44. TERRAIN-<br>SEA STATE   |          |        |  |         |   |          | FOR RUNS BELOW 5,000 FEET  A-DESERT E-LAKES/RIVERS B-FLATLAND F-MOUNTAINS C-UMG E M-WODED                                    |
| 45. INTERCEPT<br>RESULTS  |          |        |  |         |   |          | USE CODES ON FOLLOWING LINE ALL CDOES OTHER THAN A, B, C, E & F-   |
| A - SATISFACTORY INT<br>B - SAT - NO ROR LOCK:<br>C - COMPLETED REATT<br>E - INCOMPLETE REAT<br>46_FMSAEG | ON ATTEN | PT G-A | ANCELED II<br>I RADAR<br>ARGET<br>V TRANSHIT | L.II    | K IR SE<br>ITERLDCKS<br>DMPUTER<br>SLOT/RIG T | NOT SAT. |  |
| 46. FWSAEG<br>INTERCEPT<br>EVALUATION<br>IIND-FMSAEG-8811/2 (   |          |        |  |         | n filled in                                   |          | THIS LINE NOT FOR SQUADRON USE   |

Figure 3-1. Air-to-Air Missile Weapon System Flight Report, Type I, 11ND-FMSAEG-8811/2 (6-64) (Parts 1 and 2).

| CONFIDENT  | IAL (when filled in)                      |   |
|--|---|---|
| F  | RING DATA                                 |   |
| 47. INTERCEPT<br>RUN NUMBER                      |   | E IN APPLICABLE INTERCE   |
| 48. LAUNCHER<br>STATION CODE                     | WRIT<br>COOI<br>(COL                      | E IN LAUNCHER STATION<br>FROM MISSILE 10 SECTION  |
| 49. TYPE FIRING                                  | B - A                                     | CTUAL OR ATTEMPTED<br>INGLE)<br>CTUAL OR ATTEMPTED  |
| 50. FIRING RANGE                                 | C - SI<br>REPO                            | MULATED IRT TO NEAREST TENTH IN ICAL MILES MPLE: 2,5)   |
| 51. ASPECT TO<br>BOGEY HEADING                   | REPO                                      | RT IN DEGREES PLE: TAIL ON 00 HEAD ON - 1800  |
| 52. FIGHTER<br>ALTITUDE                          |   | RT IN FEET<br>EAREST HUNOREO)   |
| 53. FIGHTER<br>MACH                              | REPO                                      | RT IN NEAREST HUNOREOTI   |
| 54. FIGHTER<br>MANEUVER                          | B - PU<br>C - SN                          | RMAL TURN F - NONE<br>ISH OVER<br>AP UP   |
| 55. BOGEY<br>ALTITUOE                            | REPO                                      | GH 'G' TURN<br>RT IN FEET<br>EAREST HUNORED)  |
| 56. BOGEY MACH                                   | REPO!                                     | PLE: 1.63 OR 0.85)  |
| 17. MISS DISTANCE 8. FUZE OR CARREAD             | OR ES'<br>A - OIS<br>6 - BA<br>A - NO     |   |
| S. LAUNCH<br>ICTION                              | X - 01                                    | HER (EXPLAIN-GIVE TIME OISTANCE FROM LAUNCH) RMAL F. JETTISON DOENTAL FIRE (missile expended) FIRE (missile not e-pended) |
| D. HOW WAS<br>HSSILE<br>LSSESSED                 | 8: 51                                     | BALL<br>EMETRY<br>TER (EXPLAIN)   |
| 1. FMSAEG<br>VALUATION                           | THIS S                                    | PACE NOT FOR  |
|  | II SUPPLEMENT                             |   |
| 2. MISSILE<br>10 OE                              | 8 - WID                                   | ER (EXPLAIN)  |
| 3. FIRING<br>ROCEOURE                            | B · BOR                                   | MAL FIRING<br>ESIGHT<br>ERLOCKS OUT (OVERRIOE)<br>UAL TRACK   |
| I. WEATHER<br>T FIRING                           | A - CAV<br>X - OTH                        | U<br>ER (EXPLAIN)   |
| 5. BACKGROUND                                    | SUPPLEMENT                                |   |
| FFECTING   | A - NON<br>B - CLO<br>C - SUN<br>X - OTHI | ER (EXPLAIN)  |
| MAL FUNCTION AP                                  | ALYSIS - POST FLIGHT                      |   |
| i. MISSILE<br>AL FUNCTION                        |   |   |
| . LAUNCHER<br>ALFUNCTION                         | SHEET A                                   | ES FROM AAMREP COOE<br>NO EXPLAIN (IN REMARKS)  |
| . AM C S<br>ALFUNCTION<br>ID-FMSAEG-8811/2/6-64) | AS REQU                                   | IREO  |

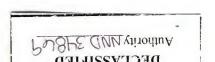
| LOCATION 150 TO THE CONTROL UNIT PRESENTAL NUMBER 150 TO THE CONTROL SERIAL NUMBER 150 TO THE CONTR | NCHER<br>TTLE<br>SSURE<br>W · 1C) |
|--|-----------------------------------|
| STATION  STATION  CONTROL SERIAL NUMBER  CONTROL SERIAL NUMBER  CONTROL SERIAL NUMBER  SERIAL NUMBER  SERIAL NUMBER  CONTROL SERIAL NUMBER  SERIAL NUMBER  CONTROL SERIAL NUMBER  SERIAL NUMBER  CONTROL SERIAL NUMBER   | SSURE<br>W - 1C)                  |
| CONTROL SERIAL NUMBER  POR G R  A DUTSONADO OR SINGLE  PORT WING INBOARD OR SINGLE  PORT WING INBOARD  PORT FUSELAGE FUSELAGE FUSELAGE FUSELAGE FUSELAGE AFT.  STAD FUSELAGE SINGLE SING |                                   |
| A DUTBOARD OR SINGLE  B PORT WING INBOARD  POBT FUSELAGE FORWARD OR SINGLE  TUSELAGE FUSELAGE FUSELAGE FUSELAGE AFT.  STBO FUSELAGE AFT.  STBO SINGLE  STBO WING INBOARD  STBOARD  STBO | 79, LANDING                       |
| BBOARO BORNARO |                                   |
| FUSELAGE FORWARD OR SINGLE FORWARD OR SINGLE FUSELAGE AFT. STRED FUSELAGE AFT. STRED FUSELAGE FUSELAGE AFT. STRED FUSELAGE FUSELAGE STRED FUSELAGE STRED STR | _                                 |
| FUSELACE AFT.  UTAD FUSELAGE FUSELAGE FUSELAGE FUSELAGE FUSELAGE FUSELAGE FUSELAGE SINGLE  SINGLE  STED WING UNTSOAROOR SINGLE  STED WING SINGLE STED WING SINGLE STED WING SINGLE STED WING SINGLE STED WING SINGLE STED WING SINGLE STED WING SINGLE STED WING SINGLE STED WING ST |                                   |
| FUSELAGE AFT.  STBO STBO STBO STBO STBO STBO STBO STB  |                                   |
| FUSELAGE FRUE, OR STROLLE STROW WING INBOARD STROW WING OUTBOARD OR SINGLE   |                                   |
| INBOARO STBO UNIG DUTBOARO OR SINGLE Use codes from AAMREP Code Sheet  |                                   |
| IOUTBOARD OR<br>SINGLE   |                                   |
| Secodes from AAMREP Code Sheet  MARKS: Use this space for necessary explanations as indicated by instructions, coding requirements and in instances  cading seems inadequals.  |                                   |
|  | es where                          |

Figure 3-1. Air-to-Air Missile Weapon System Flight Report, Type I, 11ND-FMSAEG-8811/2 (6-64) (Parts 3 and 4).

| CORC              | NA, C | LIFORMIA   | 91720     |        | EFING E  |       | _       |         |                        |         |                            |
|-------------------|-------|------------|-----------|--------|----------|-------|---------|---------|------------------------|---------|----------------------------|
| AIRCRAFT          |       | LOT        | CAL       |        | EFING L  | AIA   | _       |         | CAP                    |         |                            |
| AIRCRAFT          | -     | 201        | CAL       |        |          |       |         |         |                        |         |                            |
|                   |       |            |           | -      |          | IF    | F       |         | FREQ. C                | HANNELS |                            |
|                   |       |            |           | 1      |          |       | 111.    |         |                        |         |                            |
|                   |       |            |           | _      |          |       |         |         | PRESSUE                | E BATIC |                            |
|                   |       |            |           | ľ      | LTIMETER |       |         |         | PRESSUR                | E KAIIC |                            |
|                   | -     |            | -         | -      | EATHER   |       |         |         |                        |         |                            |
|                   |       |            |           |        |          |       |         |         |                        |         |                            |
|                   |       |            | D         | VERT   | IELD IN  | FORM  | TION    |         |                        |         |                            |
| FIELD             |       | BE         | ARING     |        | OISTAN   | Œ     |         | TAC     | AN                     |         | COMM                       |
|                   |       |            |           |        |          |       |         |         |                        |         |                            |
|                   |       |            |           |        |          |       | +       |         |                        | +-      |                            |
|                   |       |            |           |        |          |       |         |         |                        |         |                            |
|                   |       | -          |           | +      |          |       | +       |         |                        | +       |                            |
|                   |       |            |           |        |          |       |         |         |                        |         |                            |
| FLIGHT TI         | ME    |            |           |        | MARS     | HALIN | G INF   | ORMAT   | ION                    |         |                            |
| TAKE OFF          |       | 112        |           | BE     | RING     | D     | ISTAN   | 33      | ALTITE                 | DE      | E. A. C.                   |
|                   |       | MARSHAL    |           |        |          |       |         |         |                        |         |                            |
| LANDING           |       |            |           |        |          | +-    |         | -       |                        |         |                            |
|                   |       | MARSHAL    |           |        |          |       |         |         |                        |         |                            |
| BRIEFING NOTE:    | ie .  |            |           |        |          | _     |         |         |                        |         |                            |
|                   |       |            |           |        |          |       |         |         |                        |         |                            |
|                   |       |            |           |        |          |       |         |         |                        |         |                            |
|                   |       |            |           |        |          |       | _       |         |                        |         |                            |
| I, NAVY/MARINE    | SOON  | 1 2 SHIP/C | TATION (T |        | LOCATIO  |       |         | 3. OATE | (MO, DAY,              | YR)     | 4. EVENT NO.               |
| 1, 100 17 MARINE  | 3404. | 2. 3.1173  | TATION (T |        | 2021111  |       |         |         |                        |         |                            |
| S. SYLL ABUS CODE | CODE  | EX 7, AIR  | CRAFT TYP | E 8.   | AST 5 DI | GITS) | COD     | GEAR I  | O. CONTROL<br>CALL SIG | LER     | 11. CONTROL TYPE<br>(CODE) |
|                   | MIS   | SION INFO  | RMATION ( | CODE)  |          | -     | IS. PIL | TIALS)  | 16. FLIGH              | TIME    | 7. A/C LANDING             |
|                   |       | 13. COMPL  |           | 14. RE | ASON     |       | (3 INI  | ((ALS)  | HOURS                  | TENTHS  | CONDITION (CODE            |
| 12. PROPOSED      |       | IST        | 2ND       | IST    | 2ND      |       |         |         |                        |         |                            |
| 12. PROPOSED      |       |            |           |        |          |       |         |         |                        |         |                            |

|   |                                      | norm-facts | DNFIDEN   | ERCEPT  | -  | -                      |   |
|---|--------------------------------------|------------|---|---------|--|------------------------|---|
| 18. INTERCEPT<br>RUN NUMBER   | 1                                    | 2          | 3   | 4       | 5  | 6                      | INSTRUCTIONS AND COOL   |
| 19. FIGHTER<br>ALTITUDE   |                                      |            |   |         | -  |                        | REPORT IN FEET<br>(TO REAREST HUNDRED)  |
| 20. FIGHTER<br>MACH   |                                      |            |   |         |  |                        | REPORT TO NEAREST HUNGRE  |
| 21. TERRAIN-<br>SEA STATE   |                                      |            |   |         |  |                        | FOR RUNS BELOW 5,000 FEET<br>A-OESERT E-LAKES RIV<br>B-FLATLAND F-MOUNTAIN<br>C-JUNGLE H-WOODED<br>1, 2, 3, 4, 5, 6 - SEA STATE |
| 22. WEATHER AT<br>FIGHTER   |                                      |            |   |         |  |                        | A . CAVU E . SNOW<br>B . CLOUOS F . THUNDER<br>C . RAIN X . OTHERIE   |
| 23. CONTROLLER<br>NUMBER  |                                      |            |   |         |  |                        | GIVE APPROPRIATE IDENTIFIC<br>NUMBER ASSIGNED TO CONTRO<br>A - NO CONTROLLER  |
| 24. ASSESSMENT<br>OF CONTROLLER   |                                      |            |   |         |  |                        | A - NO ERROR O-BEARING ERR - DEGREES (FX F - ALTITUDE ERR - FEET (EX 1 M - RANGE ERR MILES (EX M2)                              |
| 25. RAOAR MODE  |                                      |            |   |         |  |                        | E-IR P-VISUAL G-ARO R-NORMAL H-HOJ X-OTHER(E  |
| 26 BOGEY<br>ASPECT ANGLE  |                                      |            |   |         |  |                        | REPORT IN DEGREES   |
| 27. DETECTION<br>OPPORTUNITY<br>RANGE   |                                      |            |   |         |  |                        | REPORT IN NAUTICAL MILES  |
| 28. BOGEY<br>ALTITUDE   |                                      |            |   |         |  |                        | REPORT IN FEET<br>(TO NEAREST HUNORED)  |
| 29. BOGEY MACH  |                                      |            |   |         |  |                        | REPORT TO NEAREST HUNDRE<br>(EXAMPLE: 1.21 OR 0.85)<br>REPORT Y <sub>C</sub> IF MACH IS UNKNO<br>(EXAMPLE: 1200 KNOTS)          |
| 30. DETECTION<br>RANGE  |                                      |            |   |         |  |                        | REPORT IN NAUTICAL MILES  |
| 31. LOCK-ON<br>RANGE  |                                      |            |   |         |  |                        | REPORT IN NAUTICAL MILES  |
| 32. WEATHER<br>AT BOGEY   |                                      |            |   |         |  |                        | A - CAVU E - SNOW<br>B - CLOUDS F - THUNDERS<br>C - RAIN X - OTHER (EX  |
| 33. WAS CM<br>ENCOUNTERED   |                                      |            |   |         |  |                        | A-YES B-NO IF YES, FILL OUT SUPPLEMEN CARD AND ATTACH TO THIS F   |
| 34. BOGEY TYPE  |                                      |            |   |         |  |                        | USE COOES BELOW   |
| 35. BOGEY<br>CHARACTERISTICS  |                                      |            |   |         |  |                        | A - AUGMENTED-NOT MANEUVE<br>B - AUGMENTED-MANEUVERED<br>C - UNAUGMENTED-NOT MANEU<br>E - UNAUGMENTE O-MANEUVER                 |
| 36. INTERCEPT<br>RUN RESULTS  |                                      |            |   |         |  |                        | USE CODES BELOW, ALL CODE<br>OTHER THAN A, B, AND F INDI<br>TROUBLE AND MUST BE EXPL<br>IN REMARKS.                             |
| CODES   | FOR ITEM 3                           | 34 - BO    | GEY TYPE  |         | CODES F  | FOR ITEN               | A 36 - INTERCEPT RUN RES  |
| AIRCRAFT  | TYPES                                | ОТН        | HER THAN A  | C TYPES | A SATISFA  | ACTORY IN              | TERCEPT T- AIR CONTROL<br>K ATTEMPT W.INTERMITTENT  |
| A-A4<br>B-F4B, F4C<br>C-F2B<br>E-F8<br>F-F101, F104, F105<br>G-F102, F106<br>H-B52, KC135, 707, D0<br>K-ZA6A<br>L-EF10B<br>P-E1B, E2A | R-A3A<br>T-A5<br>X-OTHER<br>(EXPLAIN | N)         | 2 - DEL MAR<br>3 - AQM-37A<br>4 - QF9F<br>5 - POGO<br>6 - HYAR<br>7 - AQM-34<br>8 - PARA FLA<br>9 - BQM-34A<br>X - OTHER (E |         | F. CANCEL<br>G. AI RADA<br>H. TARGET<br>K. IR SET<br>L. INTERLI<br>M. COMPUT<br>P. PILOT T | L ED BY COM<br>AR<br>T | MMAND RAGAR LOCK  x-OTHER (EXPL A)  SAT.  |
| 37. FMSAEG<br>INTERCEPT<br>EVALUATION   |                                      |            |   |         |  |                        | THIS LINE NOT FOR SQUADRO   |
| 11ND-FMSAE G-8811/4   | (12-64)                              | -07        | ONCIDE  | NTIAL   | (when fill-  | ad in)                 | PAGE  |

Figure 3-2. Air-to-Air Missile Weapon System Flight Report, Type II, 11ND-FMSAEG-8811/4 (12-64) (Parts 1 and 2).



|                                  |         |     | F       | IRING D | ATA       |          |   |
|----------------------------------|---------|-----|---------|---------|-----------|----------|---|
| 38 INTERCEPT                     |         |     |         |         |           |          | WRITE IN APPLICABLE INTERCEPT<br>RUN NUMBER                                   |
| RUN NUMBER                       |         |     |         |         |           |          | RUN NUMBER  |
| 39. LAUNCHER                     |         |     |         |         |           |          | WRITE IN LAUNCNER STATION CODE  |
| STATION CODE                     |         |     |         |         |           |          | FROM MISSILE ID SECTION (COL. 57)   |
| 40, TYPE OF                      |         |     |         |         |           |          | B. ACTUAL OR ATTEMPTED (SINGLE)   |
| FIRING                           |         |     |         |         |           |          | C - SIMULATED   |
|                                  |         |     |         |         |           |          | REPORT TO NEAREST TENTN IN  |
| 41. FIRING RANGE                 |         |     |         |         | •         | •        | (EXAMPLE: 2.5)  |
| 42.ASPECT TO<br>BOGEY<br>HEADING |         |     |         |         |           |          | REPORT IN DEGREES<br>EXAMPLE: TAIL ON - 00<br>HEAD DN - 1800                  |
| 43. FIGHTER                      |         |     |         |         |           |          | REPORT IN FEET  |
| ALTITUDE                         |         |     |         |         |           |          | (TO NEAREST HUNDRED)  |
| 44. FIGHTER                      |         |     |         |         |           |          | REPORT TO NEAREST HUNDREDTH   |
| MACH MACH                        |         |     |         |         |           |          | (EXAMPLE: 1.63 OR 0.85)   |
| 45. FIGHTER                      | •       | •   | -       |         |           |          | A - NORMAL TURN F - NONE<br>B - PUSH OVER                                     |
| MANEUVER                         |         |     |         |         |           |          | C - SNAP UP<br>E - NIGH 'G' TURN  |
|                                  |         |     |         |         |           |          | REPORT IN FEET  |
| 46. BOGEY                        |         |     |         |         |           |          | (TO NEAREST HUNDRED)  |
| 47. BDGEY MACH                   |         |     |         |         |           |          | REPORT TO NEAREST NUNDREOTH<br>(EXAMPLE: 1,63 OR 0.85)                        |
| 48. MISS                         | -       |     | -       | 1       |           |          | REPORT IN FEET (TELEMETERED OF<br>ESTIMATED) OR USE CODES                     |
| 01STANCE                         |         |     |         |         |           |          | A-DIRECT HIT X-OTHER (explain<br>B-BALLISTIC                                  |
| 49. FUZE/WAR -                   |         | -   | +       | 1       |           |          | A - NORMAL<br>B - NONE  |
| HEAD ACTION                      |         |     |         |         |           |          | X - OTNER (EXPLAIN-GIVE TIME OR<br>DISTANCE FROM LAUNCN)                      |
| 50, LAUNCH                       |         | -   | _       |         |           |          | A · NORMAL F · JETTISON<br>B · ACCIDENTAL                                     |
| ACTION                           |         |     |         |         |           |          | B. ACCIDENTAL C. MISFIRE (MISSILE EXPENDED) E. MISFIRE (MISSILE NDT EXPENDED) |
| 51, HDW WAS                      |         | -   | +       | +       | 1         |          | A-EYEBALL<br>B-TELEMETRY  |
| MISSILE                          |         |     |         |         |           |          | X - OTHER (EXPLAIN)   |
| ASSESSED<br>52. BACKGRDUND       |         | -   | _       | 1       |           |          | A-NONE<br>B-CLOUO   |
| AFFECTING                        |         |     |         |         |           |          | C-SUN<br>X-OTHER (EXPLAIN)  |
| GUIDANCE<br>53. FMSAEG           |         | -   | -       | -       | 1         | -        |   |
|                                  |         |     |         |         |           |          | THIS LINE IS NOT FOR SQUADRON U   |
| EVALUATION                       |         | MAI | FUNCTIO | N ANAL  | YSIS - PO | STFLIGHT | Т   |
|                                  |         | T   |         | T       |           |          |   |
| 54. MISSILE                      |         |     |         |         |           |          |   |
| MALFUNCTION                      |         |     |         |         |           |          |   |
| 55, LAUNCHER                     |         |     |         |         |           |          | USE CODES FROM AAMREP CODE<br>SNEET AND EXPLAIN (IN REMARKS)                  |
| MALFUNCTION                      |         |     |         |         |           |          | AS REQUIRED.  |
| 56. AMCS                         |         |     |         |         |           |          |   |
| MALFUNCTION                      |         |     |         |         |           |          |   |
|                                  | (12-64) |     |         |         |           | -        | PAGE 3  |

|          |  |          |                                 |                     | MISSI    | LE IDENTIFICAT                         | ION DA                      | TA                            |  |
|----------|--|----------|---------------------------------|---------------------|----------|--|-----------------------------|-------------------------------|--|
|          | LAUNCHER<br>STATION                    | E stake  | BER OF                          | D TYPE<br>of right! | TYPE     | 62. GUIDANCE<br>AND CONTROL<br>SECTION | BOT<br>PRES<br>(S/W         | TLE<br>SURE<br>- 1C)          | CODES FOR COLUMNS<br>58, 59, 60 & 61   |
| 37. CODE | LOCATION                               | S8. TONE | 59. RUN NUMBER OF<br>UNSAT TONE | 60. WARHEAD         | 61. FUZE | SERIAL NUMBER<br>INDICATE<br>P OR G    | 63. TAKE-<br>0FF<br>(P 51G) | 64. LAND.<br>ING<br>(P S / G) | SB. MISSILE TONE A-Y-ES B-NO (FAILURE) C-INTERNITTENT E-YES ON DECK NO IN AIR F-TUNEO UP LATE (MK 30 MOOO) G-NOT CHECKED |
| A        | PORT WING                              |          |                                 |                     |          |  |                             |                               | N - WEAK   |
| С        | PDRT<br>FUSELAGE<br>UPPER OR<br>SINGLE |          |                                 |                     |          |  |                             |                               | 59. GIVE RUN NUMBER OR CODE NO. (BELOW) ON WHICH TONE WAS CNECKED UNSAT. 7. ALL. RUNS 8. DECK ONLY                       |
| D        | PORT<br>FUSELAGE<br>LOWER              |          |                                 |                     |          |  |                             |                               | X - OTHER (EXPLAIN)  60. WARHEAD TYPE  A - EXERCISE (SPOTTING CHARGE)  B - LIVE (EXPLOSIVE)                              |
| E        | STBD FUSELAGE<br>LOWER                 |          |                                 |                     |          |  |                             |                               | C - IN ERT<br>E - TM<br>X - OTN ER (EXPLAIN)   |
| F        | STBD<br>FUSELAGE<br>UPPER OR<br>SINGLE |          |                                 |                     |          |  |                             |                               | 61. FUZE TYPE A- MK 322 B- MK 323  |
| н        |  |          |                                 |                     |          |  |                             |                               |  |
|          |  |          |                                 |                     |          |  |                             |                               |  |
|          |  |          |                                 |                     |          |  |                             |                               |  |

Figure 3-2. Air-to-Air Missile Weapon System Flight Report, Type II, 11ND-FMSAEG-8811/4 (12-64) (Parts 3 and 4).

CODE SHEET FOR TYPE I (F4, F3, TYPE A/C)
AIR-TO-AIR MISSILE WEAPON SYSTEM FLIGHT REPORT IT IS CONSIDERED FEASIBLE FOR THE CODES ON THIS SHEET TO BE ENTERED ON THE AAMREP FORM PRIOR TO OR IMMEDIATELY FOLLOWING THE MISSION. FOR THOSE ITEMS WHICH SHOULD BE CODED DURING FLIGHT, CODES WILL APPEAR DIRECTLY ON THE FORM. NOTE: Use a dash ( - ) in all cases in which the item is not applicable. SOLIADRON DATA INTERCEPT DATA 14. MISSION PROPOSED
NON-WEAPON SYSTEM TYPE 38. BOGEY TYPE 6. TYPE OF EXERCISE AIRCRAFT TYPES OTHER THAN A/C TYPES A - A4 B - F4B, F4C 2 - DELMAR B - MISSEX 3 - AQM-37A C - LANT MISSEX WEAPON SYSTEM TYPE C - F3B 4 - QF9F A - SPARROW III LAUNCH F - WESTPAC MISSEX 5 - POGO E - F8 B - SIDEWINDER LAUNCH F - MIDPAC MISSEX F - F101, F104, F105 6 - HVAR C - SPIII + S/W LAUNCH G - EASTPAC MISSEX 7 - AQM-34 E - AIR-TO-AIR SUPERIORITY G - F102, F106 H - MED MISSEX H - B52, KC135, 707, DC8 8 - PARA FLARE K - PREWEP TRAEX F - COMBAT AIR PATROL G - ESCORT FIGHTER K - FA6A 9 - BQM-34A L - WEPTRAEX X - OTHER (EXPLAIN) L - EF10B P - COMPEX H - INTERCEPT TRAINING K - SPECIAL WEAPONS P - E1B, E2A R - NORMAL TRAINING L - RESEARCH AND DEVELOPMENT R - A3A T - O. R. I. P - SCRAMBLE X - OTHER (EXPLAIN) - CM FLIGHT TEST X - OTHER (EXPLAIN) 9. AIRCRAFT TANK CONFIGURATION X - OTHER (EXPLAIN) 1 - ONE TANK 15. MISSION COMPLETION 2 - TWO TANKS A - COMPLETED 3 - THREE TANKS - CLEAN (NO TANKS) B - PARTIALLY COMPLETED C - NOTHING COMPLETED X - OTHER (EXPLAIN) E - AIRCRAFT NOT LAUNCHED MALFUNCTION ANALYSIS - POST FLIGHT 10. FLIGHT GEAR 16. REASON FOR DEGREE COMPLETED A - NORMAL 67. LAUNCHER MALFUNCTION 66. MISSILE MALFUNCTION B - EXPOSURE SUIT A - SAT COMPLETION A - CHECKS SAT B - IGNITER SAFE/ARM B - TM OR INSTRUMENT TROUBLE A - CHECKS SAT C - PRESSURE SULT B - MOTOR FIRE LEAD C - MISSILE TROUBLE E - LAUNCHER TROUBLE NOT ARMED (SP III) IMPROPERLY CONNECTED (SP III) 12. TYPE OF CONTROLLER C - SAFETY PIN NOT REMOVED C - UMBILICAL NOT RETRACTED A - MISSILE RANGE - RADAR F - CANCELLED BY COMMAND G - AI RADAR TROUBLE (SP III) B - SELF CONTROLLED E - LAUNCHER CHECKS UNSAT H - TARGET TROUBLE CHECKS UNSAT (USE FOLLOWING CODES AND EXPLAIN TYPE OF (EXPLAIN) J - CW TRANSMITTER TROUBLE E - GCI K - IR SET TROUBLE TEST EQUIPMENT USED) T - MULTIPLE (EXPLAIN) X - OTHER (EXPLAIN) M - COMPUTER TROUBLE 68. AMCS MALFUNCTION E - HEAD HYDRAULICS P = RADIO AND COMMUNICATION TROUBLE
R = A/C TROUBLE (OTHER THAN AMCS) STREAMED (SPIII) A - CHECKS SAT B - VERTICAL GYRO DATA LINK F - EPU/BATTERY FIRED T - AIR CONTROL TROUBLE F - ATDS G - A! RADAR TROUBLE G - SAGE X - OTHER (EXPLAIN) G - HPIL FIRED (SP III) J - CW ILLUMINATOR H - MTDS K - IR SET TROUBLE H - GAS GRAIN FIRED (S/W) 20. AIRCRAFT LANDING CONDITION J - NTDS K - NITROGEN PRESSURE M - COMPUTER TROUBLE A - UP TROUBLE (S/W-1C) T - MULTIPLE (EXPLAIN) DOWN BECAUSE OF: VOICE T - MULTIPLE (EXPLAIN) X - OTHER (EXPLAIN) K - MTDS B - AIRFRAME X - OTHER (EXPLAIN) C - ORDNANCE L - NTDS E - ELECTRONICS F - POWER PLANT P - ATDS R - WF-2 T - MULTIPLE (EXPLAIN) X - OTHER (EXPLAIN) 13. BIT CHECK MISSILE IDENTIFICATION DATA AA - SAT PERFORMANCE BB - NOT PERFORMED 70. GIVE RUN NUMBER OR CODE NUMBER (BELOW) ON WHICH TONE WAS UNSAT BITS CHECKED UNSAT. MA - BIT ZERO - B+ 7 - ALL RUNS 88 - BIT ZERO - DISPLAY - DECK ONLY IC - BIT ONE - DETECTION ID - BIT ONE - ACQUISITION X - OTHER (EXPLAIN) 1E - BIT ONE - COMP 2F - BIT TWO - RANGE TRACK 71. WARHEAD TYPE - (SPARROW III AND SIDEWINDER) A - EXERCISE (SPOTTING CHARGE) 2G - BIT TWO - COMP B - LIVE (EXPLOSIVE) C - INERT 3H - BIT THREE - ANGLE TRACK 4J - BIT FOUR - COMP AK - BIT FOUR - ADJ/HOJ X - OTHER (EXPLAIN) 5L - BIT FIVE - COMP 6M - BIT SIX - COMP 72, 73, & 75. SPARROW III SELECT LIGHT AND READY LIGHT AND 7W - BIT SEVEN - IR SIDEWINDER TONE 9X - MULTIPLE UNSAT BITS (EXPLAIN) A - YES B - NO (FAILURE) C - INTERMITTENT E - YES ON DECK NO IN AIR F - TUNED UP LATE G - NOT CHECKED X - OTHER (EXPLAIN) 76. FUZE TYPE A - MK 322 B - MK 323 11ND-FMSAEG-8811/3 (6-64)

Figure 3-3. Form 11ND-FMSAEG-8811/3 (6-64).

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### CODE SHEET FOR TYPE II (F8, A4, A6 TYPE A/C) AIR-TO-AIR WEAPON SYSTEM FLIGHT REPORT

11ND-FMSAEG-8811/5 (6-64)

H - GAS GRAIN FIRED

T - MULTIPLE (EXPLAIN)

X - OTHER (EXPLAIN)

(S/W - 1C)

K - NITROGEN PRESSURE TROUBLE

CODES ON THIS SHEET MAY BE ENTERED ON THE AAMREP FORM PRIOR TO OR IMMEDIATELY FOLLOWING THE FLIGHT. OTHER CODES WILL APPEAR DIRECTLY ON THE AAMREP FORM.

NOTE: Use a dash ( - ) in all cases in which the item is not applicable.

#### SQUADRON DATA 14. REASON FOR DEGREE OF COMPLETION 6. TYPE OF EXERCISE 12. MISSION PROPOSED A - SAT COMPLETION A - AAWEX NON-WEAPON SYSTEM TYPE B - TM OR INSTRUMENT TROUBLE 3 - ALL NON-WEAPON TYPE B - MISSEX C - MISSILE TROUBLE C - LANT MISSEX E - LAUNCHER TROUBLE WEAPON SYSTEM TYPE E - WESTPAC MISSEX A - SPARROW III LAUNCH F - CANCELLED BY COMMAND F - MIDPAC MISSEX B - SIDEWINDER LAUNCH G - AI RADAR TROUBLE G - EASTPAC MISSEX C - SP III + S/W LAUNCH H - TARGET TROUBLE H - MED MISSEX J - CW TRANSMITTER TROUBLE E - AIR-TO-AIR SUPERIORITY K - PREWEP TRAEX K - IR SET TROUBLE F - COMBAT AIR PATROL L - WEPTRAEX M - COMPUTER TROUBLE G - ESCORT FIGHTER P - COMPEX P - RADIO & COMMUNICATION TROUBLE H - INTERCEPT TRAINING R - NORMAL TRAINING R - A/C TROUBLE (OTHER THAN AMCS) K - SPECIAL WEAPONS T - O. R. I L - RESEARCH AND DEVELOPMENT T - AIR CONTROL TROUBLE X - OTHER (EXPLAIN) X - OTHER (EXPLAIN) P - SCRAMBLE R - CM FLIGHT TEST 9. FLIGHT GEAR 17. AIRCRAFT LANDING CONDITION X - OTHER (EXPLAIN) A - NORMAL A - UP B - EXPOSURE SUIT 13. MISSION COMPLETION C . PRESSURE SUIT DOWN BECAUSE: A - COMPLETED B - PARTIALLY COMPLETED B - AIRFRAME 11. TYPE OF CONTROLLER C - ORDNANCE C - NOTHING COMPLETED A - MISSILE RANGE - RADAR E - AIRCRAFT NOT LAUNCHED E - ELECTRONICS B - SELF CONTROLLED F - POWER PLANT C - CIC T - MULTIPLE (EXPLAIN) E - GCI X - OTHER (EXPLAIN) T - MULTIPLE (EXPLAIN) X - OTHER (EXPLAIN) DATA LINK F - ATDS G - SAGE H . MTDS J - NTDS VOICE K - MTDS I - NTDS M - SAGE P - ATDS R - WF-2 MAL FUNCTION ANALYSIS - POSTFLIGHT 56. AMCS 55. LAUNCHER 54. MISSILE A - CHECKS SAT A - CHECKS SAT A - CHECKS SAT

Figure 3-4. Form 11ND-FMSAEG-8811/5 (6-64).

E - CHECKS UNSAT (EXPLAIN)

G - AI RADAR TROUBLE

M - COMPUTER TROUBLE

T - MULTIPLE (EXPLAIN)

X - OTHER (EXPLAIN)

K - !R SET TROUBLE

#### Appendix A

# MEASURES FOR CORRECTING DETENT WRENCH ROTATION

Some reports have indicated that the detent wrench on the LAU-7/A launcher has been rotated after the missile was secured to the aircraft. If this condition is prevalent, the following corrective measures are recommended:

- 1. Shorten detent wrench handle as follows:
- a. Remove 4 1/2 inches from end of detent wrench handle.
- b. Relocate flag on remaining portion of handle.
- c. Use crescent wrench to turn shortened wrench handle when required in missile assembly to launcher.
  - 2. Mark launcher as follows:
- a. Scribe a vertical line on each side of launcher in line with center of front striker point.
  - b. Paint a 2-inch-long vertical

black stripe 3/8 inch wide from scribe line forward on each side of launcher so scribed line becomes aft edge of black stripe.

- 3. Load missile on launcher as follows:
- a. Insert motor hangers into launcher loading slots.
- b. Raise detent by rotating detent wrench—safety pin and slide missile forward until front edge of hanger block is within the black stripe on launcher.
- c. Lower detent by releasing, but not removing, detent wrench—safety pin.
- d. Slide missile forward engaging detent. When engaged, the detent wrench rotates counterclockwise.
- e. Complete missile assembly as given in paragraph 2-1.2.

# Appendix B CHECK LISTS

|    | J101.  b. With detent raised, cable assembly lug handle inserted in position on launcher rail.  c. With launcher nose cover raised, P103 connected to launcher  | <ul> <li>B-2 AN/ASM-11 JETTISON TEST OF AERO 3A LAUNCHER</li> <li>1. All ordnance (both internal and external) and other external stores removed from aircraft.</li> <li>2. Aircraft circuitry energized.</li> </ul> |  |
|----|---|--|--|
|    | NOTE: With test set in position, knife edge pointer in extreme left edge of BLACK area.  a. Cable assembly P102 connected to test set   | <ul> <li>k. With safety pin removed, switch through positions 16, 17, and 18. BLACK.</li> <li>1. Switch to position 19. YELLOW.</li> <li>m. Switch OFF; safety pin reinserted.</li> </ul>                            |  |
|    | <ul> <li>a. Forward snubber cams have no hairline cracks on outer edges.</li> <li>b. With detent raised, free movement.</li> <li>c. Unloading stirrup secured properly and no dents or any deformation to stirrup.</li> <li>d. Safety pin installed.</li> <li>2. Aircraft checked:</li> <li>a. All armament (both internal and external) removed.</li> <li>3. Test procedures:</li> </ul> | and SAFETY OVER- RIDE switches ON.  h. Firing switch depressed,  |  |
| AE | 1 AN/ASM-11 CHECKOUT OF<br>FRO 3A LAUNCHER  Ref: NAVWEPS OP 2309, Vo 4, Third Revision, paragraph 1-2.2 through 1-2.4,  Tools: 3/8-inch hex wrench 1, Launcher checked;   | <ul> <li>e. Switch turned through first 14 positions. YELLOW in each position.</li> <li>f. On position 10, 400-cycle tone heard in pilot's earphones.</li> <li>g. MASTER ARMAMENT</li> </ul>                         |  |

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| 5. Switch turned through positions 16, 17, and 18, BLACK for each position.   | b. Aircraft engines OFF.  c. Safety pin in place.  |      |
|---|--|------|
| 6. Power turned off, or removed; test set disconnected,   | d. NPA removed from motor.  e. Dust cover on power   |      |
| 3-3 AN/ASM-11 AIRCRAFT CIRCUIT<br>TEST OF AERO 3A LAUNCHER  | supply receptacle of launcher.   |      |
| NOTE: Aircraft must have connector MS 3106A-20-27S at or near launcher station.   | 2. Loading procedures:  a. With missile just outside three loading slots, missile moved to engage it in slots. |      |
| <ol> <li>Cable assembly P102 con-<br/>nected to test set J101.</li> </ol>   | b. Aft lug of detent is raised by rotating with wrench.  |      |
| 2. P105 of adapter connected to MS 3106A-20-27S.  | c. Missile pushed forward<br>3 inches; forward lug<br>engaged detent.  |      |
| <ol> <li>Cable assembly P103 con-<br/>nected to adapter P104.</li> </ol>  | d. Wrench removed from launcher.   |      |
| <ol> <li>Aircraft circuitry ener-<br/>gized with auxiliary power<br/>unit,</li> </ol>   | e. Launcher front cover un-<br>latched, dust cap removed,<br>umbilical cord and um-                            |      |
| 5. MASTER ARMAMENT and SAFETY OVERRIDE switches ON.   | bilical disconnect mech-<br>anism connected.<br>Launcher front cover   |      |
| <ol> <li>Switch turned through po-<br/>sitions 2, 3, 10. YELLOW<br/>at all positions; 400-cycle<br/>tone heard in pilot's ear-<br/>phones.</li> </ol> | f. Snubber cams locked in position; safety pin still in place.   |      |
| <ul><li>7. Switch turned to position 12.</li><li>8. Firing switch depressed. RED on position 12.</li></ul>  | g. Covers removed from G&C section and fuze (just before aircraft moves forward).                              |      |
| <ol><li>Power turned OFF; test<br/>set disconnected.</li></ol>  | 3. Simplified missile checkout made:   |      |
| B-4 LOADING MISSILE ON<br>AERO 3A LAUNCHER  | <ul> <li>a. Power applied through<br/>aircraft circuits.</li> </ul>  |      |
| Ref: NAVWEPS OP 2309,<br>Volume 4, Third Revision,  | b. Gyro is rotating.   | Ш    |
| paragraph 2-1.1. Tools: 3/8-inch hex wrench; two-cell flashlight.   | c. With gyro rotating, two-<br>cell flashlight with glass<br>lens passed in front of<br>seeker head. Pilot's   | LJ   |
| 1. Preliminary checks:  | tone heard.  | لــا |
| <ul> <li>Aircraft battery and<br/>MASTER ARMAMENT<br/>switches OFF.</li> </ul>  | 4. Launcher safety pin pulled (aircraft taxies forward).   |      |

|             | N/ASM-11 CHECKOUT OF<br>1/A LAUNCHER  |   | 11. Firing switch released;<br>MASTER ARMAMENT<br>switch OFF.  |   |
|-------------|---|---|--|---|
| 1           | NOTE: Part No. 1001-<br>517359, FSN VM-5935-<br>85-9397-M558 is required.<br>This checkout applies to |   | 12. Safety pin removed,<br>switch turned to position<br>15. BLACK, and pilot<br>light goes out.              |   |
| ]<br>]<br>a | AU-7/A launchers with HVAR receptacle removed and wiring modified.                                    |   | 13. Safety pin removed,<br>switch turned through<br>positions 16, 17, and 18.<br>BLACK.                      |   |
| 1.          | Test set in position; knife edge pointer in extreme left edge of BLACK area.                          |   | 14. Switch turned to position 19. YELLOW.  |   |
| 2.          | Cable assembly P102 connected to test set J101.   |   | <ol><li>Switch turned OFF; safety<br/>pin reinserted.</li></ol>  |   |
| 3.          | Safety pin installed in launcher.   |   | B-6 AN/ASM-11 JETTISON TEST OF   |   |
| 4.          | With detent raised, cable   |   | LAU-7/A LAUNCHER   |   |
|             | assembly lug handle inserted in position on launcher rail.  |   | <ol> <li>All ordnance (both internal<br/>and external) and other ord-</li> </ol>                             |   |
| 5.          | With launcher nose cover raised, Pl03 connected to launcher connector (AN-                            |   | <ul><li>nance stores removed from aircraft.</li><li>2. Aircraft circuitry energized.</li></ul>               |   |
| 6.          | 3102E-22-14P).  Auxiliary power unit connected; aircraft circuitry                                    | _ | <ol> <li>Jettison test for applicable<br/>aircraft performed. Knife<br/>edge pointer in RED area.</li> </ol> |   |
|             | energized.  |   | 4. Jettison circuitry de-  |   |
| 7.          | Switch turned through<br>first 14 positions. YELLOW<br>in each position.                              |   | energized. 5. Switch turned through po-  | Ш |
| 8.          | On position 10, 400-cycle   |   | sitions 16, 17, and 19. BLACK for each position.  6. Power turned off, or removed; test set discon-          |   |
|             | tone heard in pilot's ear-<br>phones.   |   |  |   |
| 9.          | MASTER ARMAMENT<br>and SAFETY OVER-<br>RIDE switches turned ON.                                       |   | nected.  B-7 AN/ASM-11 AIRCRAFT CIRCUIT  | ٧ |
| 10.         | Firing switch depressed:  |   | TEST OF LAU-7/A LAUNCHER   |   |
|             | a. Switch turned through positions 11 and 12. RED.  |   | NOTE: Aircraft must have<br>connector MS 3106A-20-<br>27S at or near launcher                                |   |
|             | <ul> <li>Switch turned through<br/>positions 13 and 14.</li> <li>BLACK.</li> </ul>                    |   | station.  1. Cable assembly P102 connected to test set J101.   |   |
|             | <li>c. Safety pin pulled;<br/>positions 13 and 14 read<br/>again. RED.</li>                           |   | 2. P105 of adapter connected to MS 3106A-20-27S.   |   |

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| 3. Cable assembly Pl03 connected to adapter Pl04.   | <ul> <li>c. With detent raised</li> <li>by rotating detent</li> </ul>  |
|---|--|
| <ol> <li>Aircraft circuitry ener-<br/>gized with auxiliary power<br/>unit.</li> </ol>   | wrench-safety pin, missile slid forward and front edge of hanger block engaged front detent lug.                             |
| <ol> <li>MASTER ARMAMENT and<br/>SAFETY OVERRIDE switches<br/>ON.</li> </ol>  | d. Detent lowered by re-<br>leasing, but not removing.   |
| 6. Switch turned through positions 2, 3, and 10. YELLOW at all positions; 400-cycle   | e. Missile is properly loaded and detent does  |
| tone heard in pilot's ear-<br>phones.   | not rest on firing button of missile.  |
| 7. Switch turned to position 12.  | f. Nose latch button depressed and nose fairing slid forward.  |
| <ul><li>8. Firing switch depressed.<br/>RED on position 12.</li><li>9. Power turned OFF; test<br/>set disconnected.</li></ul> | g. No damage to umbilical hook. Hook attached to missile.  |
| B-8 LOADING MISSILE ON<br>LAU-7/A LAUNCHER<br>Ref: NAVWEPS OP 2309,<br>Volume 4, Third Revision,<br>paragraph 2-1.2.          | h. Forward-receptacle dust cap removed; umbilical cable of missile connected by adapter to launcher power supply receptacle. |
| Tools: 5/16-inch hex wrench.  | <ul> <li>i. Nose fairing pushed home. L</li> <li>j. Covers removed from<br/>G&amp;C section and fuze</li> </ul>              |
| <ol> <li>Preliminary checks:</li> <li>a. Cockpit switches OFF.</li> </ol>   | (just before aircraft moves forward)   |
| b. Aircraft engines OFF.  | 3. Simplified missile checkout made:   |
| c. Auxiliary power unit NOT connected.  | a. Power applied through aircraft circuits.  |
| d. Aircraft grounded.   | b. Gyro is rotating.   |
| <ul> <li>e. Safety pin in place.</li> <li>f. NPA removed from<br/>motor of missile.</li> </ul>                                | c. With gyro rotating, two-cell flashlight with glass lens passed in front of seeker head.                                   |
| <ol><li>Loading procedures:</li></ol>   | Pilot's tone heard.  |
| <ul> <li>a. Motor hangers inser-<br/>ted into launcher<br/>loading slots.</li> </ul>  | 4. Launcher safety pin pulled (aircraft taxies forward).   |
| <ul> <li>b. Missile slid forward<br/>until it hit aft detent</li> </ul>   |  |